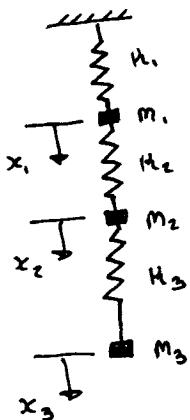


H_{ij} : the force at point i due to a unit displacement at point j . When all the other points have zero displacement.

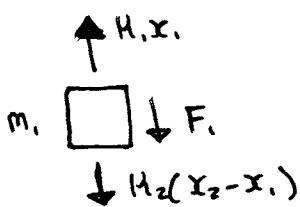
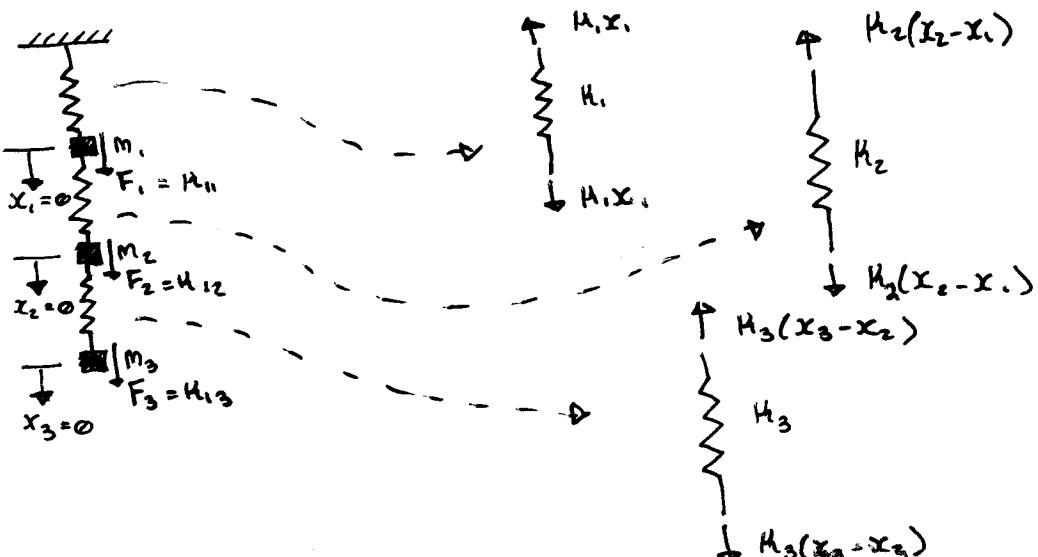
Example:



- Find the stiffness influence matrix.

$$K = \begin{bmatrix} H_{11} & H_{12} & H_{13} \\ H_{21} & H_{22} & H_{23} \\ H_{31} & H_{32} & H_{33} \end{bmatrix}$$

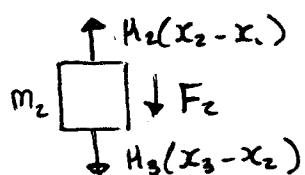
Solution: $x_1 = 0$; $x_2 = 0$; $x_3 = 0$



$$\sum F_i = 0$$

$$F_1 + H_2(x_2 - x_1) - H_1x_1 = 0$$

$$F_1 = (H_1 + H_2)x_1 - H_2x_2$$



$$\dots F_2 = -H_2x_1 + (H_2 + H_3)x_2 - H_3x_3$$

$$\dots F_3 = -H_3x_2 + H_3x_3$$

Set $X_1 = 1$; $X_2 = X_3 = 0$

$$F_1 = (H_1 + H_2)$$

$$H_{11} = H_1 + H_2$$

$$F_2 = -H_2$$

$$H_{21} = -H_2$$

$$F_3 = 0$$

$$H_{31} = 0$$

Set $X_1 = X_3 = 0$; $X_2 = 1$

$$F_1 = -H_2$$

$$H_{21} = -H_2$$

$$F_2 = H_2 + H_3$$

$$H_{22} = H_2 + H_3$$

$$F_3 = -H_2$$

$$H_{32} = -H_2$$

Set $X_1 = X_2 = 0$; $X_3 = 1$

$$F_1 = 0$$

$$H_{13} = 0$$

$$F_2 = -H_{31}$$

$$H_{23} = -H_{31}$$

$$F_3 = H_3$$

$$H_{33} = H_3$$

$$\rightarrow [K] = \begin{bmatrix} H_1 + H_2 & -H_2 & 0 \\ -H_2 & H_2 + H_3 & -H_3 \\ 0 & -H_3 & H_3 \end{bmatrix}$$

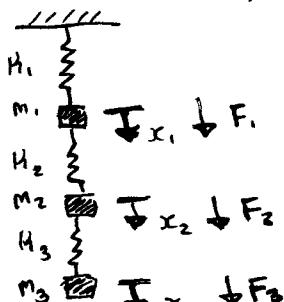
(stiffness matrix by statics)

a_{ij} : the deflection at point i due to a unit force at point j.

$$[A] = [a_{ij}] = \begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \\ \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & \dots & a_{nn} \end{bmatrix}$$

$$[A][k] = [I]$$

Example Find the flexibility influence coefficient matrix



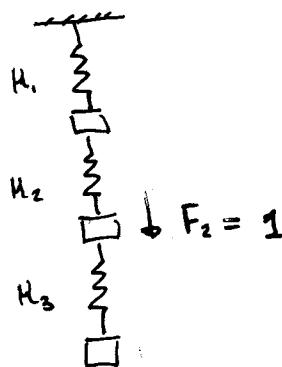
Solution:

$$F_1 = 1 ; F_2 = F_3 = 0$$

$$\begin{array}{ccc|c} \text{disp:} & X_1 & X_2 & X_3 \\ & " & " & " \\ a_{11} & a_{21} & a_{31} & \end{array}$$

Spring 1 : $F_1 = K_1 x_1 = 1 \quad (*)$
 $x_1 = 1/K_1 = x_2 = x_3$

Let $F_1 = F_3 = 0$, $F_2 = 1$



$$K_{eq} = \frac{K_1 K_2}{K_1 + K_2} \quad ; \quad x_2 = \frac{F_2}{K_{eq}}$$

$$\therefore x_2 = \frac{1}{K_1} + \frac{1}{K_2} = x_3$$

$$\alpha_{22} = x_2 = \frac{1}{K_1} + \frac{1}{K_2}$$

$$\alpha_{32} = x_3 = \frac{1}{K_1} + \frac{1}{K_2}$$

$$\boxed{\alpha_{12} = \alpha_{21} = \frac{1}{K_1}}$$

$$\frac{1}{K_{eq}} = \frac{1}{K_1} + \frac{1}{K_2} + \frac{1}{K_3}$$

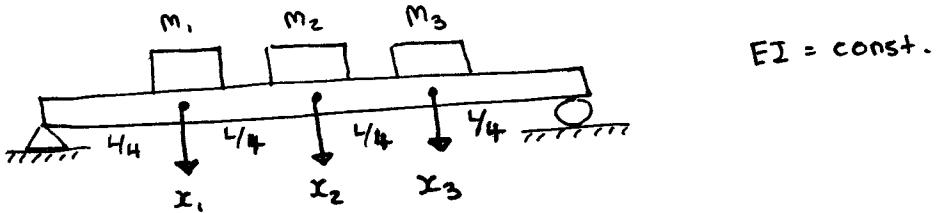
$$x_3 = \frac{F_3}{K_{eq}} = \frac{1}{K_{eq}} = \frac{1}{K_1} + \frac{1}{K_2} + \frac{1}{K_3}$$

$$\therefore \alpha_{33} = x_3 = \frac{1}{K_1} + \frac{1}{K_2} + \frac{1}{K_3}$$

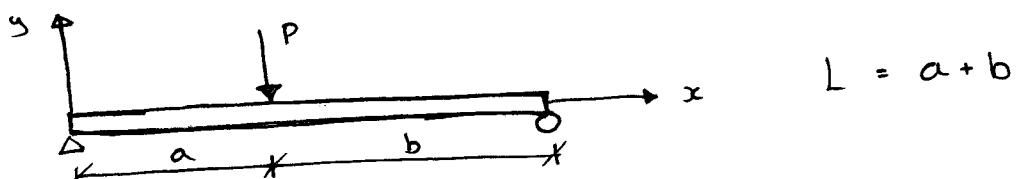
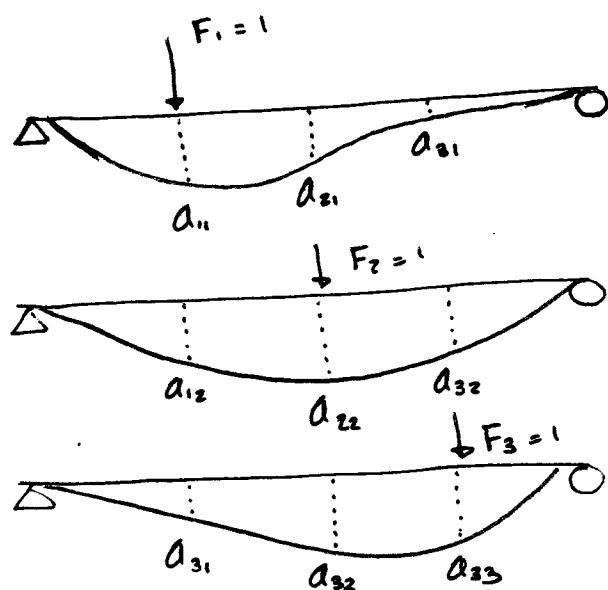
$$\alpha_{13} = \alpha_{31} = \frac{1}{K_1} \quad ; \quad \alpha_{23} = \alpha_{32} = \frac{1}{K_1} + \frac{1}{K_2}$$

$$\therefore [A] = \begin{bmatrix} 1/K_1 & 1/K_1 & 1/K_1 \\ 1/K_1 & 1/K_1 + 1/K_2 & 1/K_1 + 1/K_2 \\ 1/K_1 & 1/K_1 + 1/K_2 & 1/K_1 + 1/K_2 + 1/K_3 \end{bmatrix}$$

Example Find the flexibility matrix of the weightless beam shown:



Solution:



$$y = \begin{cases} \frac{Pbx}{6EI} (L^2 - b^2 - x^2) & ; 0 \leq x \leq a \\ -\frac{Pa(L-x)}{6EI} (a^2 + x^2 - 2Lx) & ; a \leq x \leq L \end{cases}$$

$$\alpha_{11} : a = \frac{L}{4} \quad ; \quad b = \frac{3}{4}L$$

$$\text{At } x = \frac{L}{4}$$

$$\alpha_{11} = \frac{P(\frac{3}{4}L)(\frac{1}{4}L)}{6EI} (L^2 - (\frac{3}{4}L)^2 - (\frac{1}{4}L)^2)$$

$$\alpha_{11} = \left(\frac{9}{32} \right) \left(\frac{L^3}{EI} \right)$$

At $x = L/2$

$$a_{21} = \left(\frac{11}{768}\right) \left(\frac{L^3}{EI}\right)$$

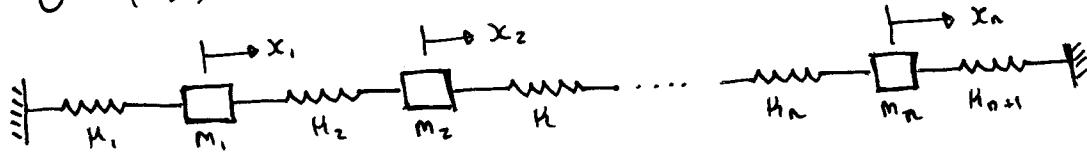
At $x = 3L/4$

$$a_{31} = \left(\frac{7}{768}\right) \left(\frac{L^3}{EI}\right)$$

$$[A] = \frac{L^3}{768EI} \begin{bmatrix} 9 & 11 & 7 \\ 11 & 16 & " \\ 7 & " & 9 \end{bmatrix}$$

Potential and Kinetic Energy :

$$U = (\frac{1}{2})Kx^2 = (\frac{1}{2})Fx$$



$$\{\vec{F}\} = [K]\{\vec{x}\}$$

$$\text{Here : } \{\vec{F}\} = (F_1, F_2, \dots, F_n)^T$$

$$\{\vec{x}\} = (x_1, x_2, \dots, x_n)^T$$

The potential energy :

$$U = (\frac{1}{2})F_1x_1 + (\frac{1}{2})F_2x_2 + \dots + (\frac{1}{2})F_nx_n$$

$$= (\frac{1}{2})(F_1, F_2, \dots, F_n) \begin{Bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{Bmatrix}$$

$$= (\frac{1}{2})\{\vec{F}\}^T \{\vec{x}\}$$

$$U = (\frac{1}{2})([K]\{\vec{x}\})^T \{\vec{x}\}$$

$$= (\frac{1}{2})\{\vec{x}\}^T [K] \{\vec{x}\}$$

↑ Stiffness matrix

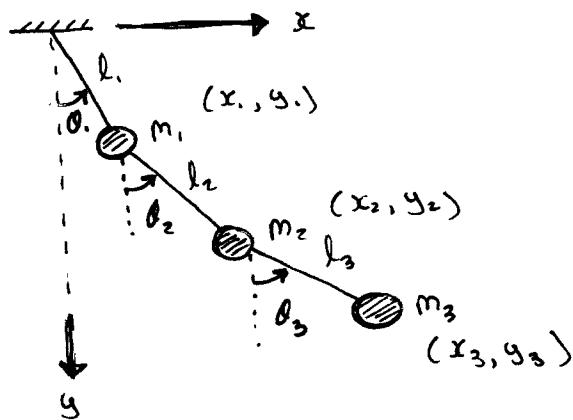
The Kinetic energy :

$$T = (\frac{1}{2})m_1\dot{x}_1^2 + (\frac{1}{2})m_2\dot{x}_2^2 + \dots + (\frac{1}{2})m_n\dot{x}_n^2$$

$$= (\frac{1}{2})\{\dot{\vec{x}}^T [M] \dot{\vec{x}}\}$$

Here $[M] = \begin{bmatrix} m_1 & 0 & \dots \\ 0 & m_2 & \dots \\ \dots & \dots & m_n \end{bmatrix}$

Generalized coordinates :



* $\theta_1, \theta_2, \theta_3$ are three independent generalized coordinates.

$$x_1^2 + y_1^2 = l_1^2$$

$$(x_2 - x_1)^2 + (y_2 - y_1)^2 = l_2^2$$

$$(x_3 - x_2)^2 + (y_3 - y_2)^2 = l_3^2$$

$$6 : x_1, y_1, x_2, y_2, x_3, y_3$$

3 constrained eqns

only $6 - 3 = 3$ are independent.

6 : $x_1, y_1, x_2, y_2, x_3, y_3$
3 constrained eqns
only $6 - 3 = 3$ are independent.

Nov. 7/19

Generalized coordinates :

$$\left\{ \begin{array}{l} q_1 = \theta_1 ; q_2 = \theta_2 ; q_3 = \theta_3 \\ x_1 = l \sin(\theta_1) ; y_1 = l \cos(\theta_1) \\ x_2 = x_1 + l_2 \sin(\theta_2) ; y_2 = y_1 + l_2 \cos(\theta_2) \\ x_3 = x_2 + l_3 \sin(\theta_3) ; y_3 = y_2 + l_3 \cos(\theta_3) \end{array} \right.$$

$$\left\{ \begin{array}{l} x_i = x_i(q_1, q_2, q_3) \\ y_i = y_i(q_1, q_2, q_3) \end{array} \right. \quad i = 1, 2, 3$$

Virtual displacement :

$$q_1, q_2, \dots, q_n \rightarrow \delta q_1, \delta q_2, \dots, \delta q_n$$

The work done : $\delta w_1, \delta w_2, \dots, \delta w_n$

The generalized Force :

$$Q_1 = \frac{\delta w_1}{\delta q_1}, Q_2 = \frac{\delta w_2}{\delta q_2}, \dots, Q_n = \frac{\delta w_n}{\delta q_n}$$

Define the Lagrangian

$$L = T - V$$

Then the equations of motion

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{q}_i} \right) - \frac{\partial T}{\partial q_i} + \frac{\partial V}{\partial q_i} = Q_i \quad ; \quad i = 1, 2, \dots, n$$

If F_{xk}, F_{yk}, F_{zk} are the external forces acting on the k^{th} mass in the x, y, z -directions,

$$Q_i = \sum_k \left[F_{xk} \left(\frac{\partial x_k}{\partial \dot{q}_i} \right) + F_{yk} \left(\frac{\partial y_k}{\partial \dot{q}_i} \right) + F_{zk} \left(\frac{\partial z_k}{\partial \dot{q}_i} \right) \right] \quad ; \quad i = 1, 2, \dots, n$$

Sudden
Changing
from i
to i , but
something

Viscously damped Systems

Rayleigh's dissipation function :

$$R = (\frac{1}{2}) \dot{x}^T [C] \dot{x}$$

here $[C]$ is the damping matrix

The equations of motion

$$[\underline{M}] \{ \ddot{x} \} + [\underline{C}] \{ \dot{x} \} + [\underline{K}] \{ x \} = \{ F \}$$

Proportional damping matrix

$$[C] = \alpha [M] + \beta [K]$$

The generalized force of the viscous damping

$$Q_i = -\frac{\partial R}{\partial \dot{x}_i}$$

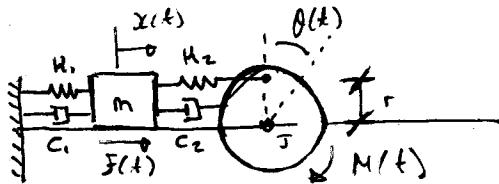
In the generalized coordinates,

$$Q_i = -\frac{\partial R}{\partial \dot{q}_i}$$

The final equations of motion:

$$\begin{aligned} \frac{d}{dt} \left(\frac{\partial T}{\partial \dot{q}_i} \right) - \frac{\partial T}{\partial q_i} + \frac{\partial V}{\partial q_i} &= -\frac{\partial R}{\partial \dot{q}_i} + Q_i \\ \Rightarrow \frac{d}{dt} \left(\frac{\partial T}{\partial \dot{q}_i} \right) - \frac{\partial T}{\partial q_i} + \frac{\partial R}{\partial \dot{q}_i} + \frac{\partial V}{\partial q_i} &= Q_i \end{aligned}$$

Example:



Derive the equations of motion.

$$\underline{Solution:} \quad q_1 = x(t) \quad ; \quad q_2 = \theta(t)$$

→ generalized coordinates

Kinetic:

$$T = (\frac{1}{2})m\dot{x}^2 + (\frac{1}{2})J\dot{\theta}^2$$

$$T = (\frac{1}{2})m\dot{q}_1^2 + (\frac{1}{2})J\dot{q}_2^2$$

Potential:

$$V = (\frac{1}{2})K_1x^2 + (\frac{1}{2})K_2(r\theta - x)^2$$

$$V = (\frac{1}{2})K_1q_1^2 + (\frac{1}{2})K_2(q_2 - \theta)^2$$

Rayleigh's dissipation function

$$R = (\frac{1}{2})C_1\dot{x}^2 + (\frac{1}{2})C_2(r\dot{\theta} - \dot{x})^2$$

$$= (\frac{1}{2})C_1\dot{q}_1^2 + (\frac{1}{2})C_2(r\dot{q}_2 - \dot{q}_1)^2$$

For \dot{q}_1 :

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{q}_1} \right) - \frac{\partial T}{\partial q_1} + \frac{\partial R}{\partial \dot{q}_1} + \frac{\partial V}{\partial \dot{q}_1} = Q_1$$

$$\rightarrow \frac{\partial T}{\partial \dot{q}_1} = m\ddot{q}_1 \quad ; \quad \rightarrow \frac{\partial T}{\partial q_1} = 0$$

$$\rightarrow \frac{\partial R}{\partial \dot{q}_1} = C_1 \dot{q}_1 + C_2 (r\dot{q}_2 - \dot{q}_1) \cdot (-1)$$

$$\rightarrow \frac{\partial V}{\partial \dot{q}_1} = H_1 q_1 + H_2 (r q_2 - q_1) \cdot (-1)$$

$$\rightarrow Q_1 = f(t)$$

$$\textcircled{1} \quad m\ddot{q}_1 + C_1 \dot{q}_1 + C_2 (r\dot{q}_2 - \dot{q}_1) + H_1 q_1 + H_2 (r q_2 - q_1) = f(t)$$

For \dot{q}_2 :

$$\rightarrow \frac{\partial T}{\partial \dot{q}_2} = J q_2 \quad ; \quad \rightarrow \frac{\partial T}{\partial q_2} = 0$$

$$\rightarrow \frac{\partial R}{\partial \dot{q}_2} = C_2 (r\dot{q}_2 - \dot{q}_1) (r) = C_2 r (r\dot{q}_2 - \dot{q}_1)$$

$$\rightarrow \frac{\partial V}{\partial \dot{q}_2} = H_2 r (r q_2 - q_1)$$

$$\rightarrow Q_2 = M(t)$$

$$\textcircled{2} \quad J\ddot{q}_2 + C_2 r (r\dot{q}_2 - \dot{q}_1) + H_2 r (r q_2 - q_1) = M(t)$$

Putting \textcircled{1} and \textcircled{2} into matrix form:

$$\begin{bmatrix} m & 0 \\ 0 & J \end{bmatrix} \begin{Bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{Bmatrix} + \begin{bmatrix} C_1 + C_2 & -C_2 r \\ -C_2 r & C_2 r^2 \end{bmatrix} \begin{Bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{Bmatrix} + \begin{bmatrix} H_1 + H_2 & -H_2 r \\ -H_2 r & H_2 r^2 \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \end{Bmatrix} = \begin{Bmatrix} f(t) \\ M(t) \end{Bmatrix}$$

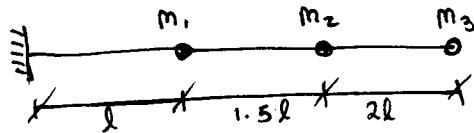
$$T = (\frac{1}{2}) m \dot{q}_1^2 + (\frac{1}{2}) J \dot{q}_2^2 = (\frac{1}{2}) (\dot{q}_1 \dot{q}_2) \begin{bmatrix} m & 0 \\ 0 & J \end{bmatrix} \begin{Bmatrix} \dot{q}_1 \\ \dot{q}_2 \end{Bmatrix}$$

$$\text{Define } \vec{q} = \begin{Bmatrix} q_1 \\ q_2 \end{Bmatrix}$$

$$T = (\frac{1}{2}) \vec{q}^T [M] \vec{q}$$

$$\begin{aligned} V &= (\frac{1}{2}) H_1 q_1^2 + (\frac{1}{2}) H_2 (r^2 q_2^2 - 2r q_1 q_2 + q_1^2) \\ &= (\frac{1}{2}) [(H_1 + H_2) q_1^2 - 2H_2 r q_1 q_2 + H_2 r^2 q_2^2] \\ &= (\frac{1}{2}) \vec{q}^T \begin{bmatrix} H_1 + H_2 & -H_2 r \\ -H_2 r & H_2 r^2 \end{bmatrix} \vec{q} \end{aligned}$$

Example



$$m_1 = 3m ; m_2 = 2m ; m_3 = m ; EI = \text{const.}$$

Find the natural frequencies and mode shapes:

Solution:

$$\begin{aligned} [A] &= \frac{l^3}{24EI} \begin{bmatrix} 8 & 26 & 50 \\ 26 & 125 & 275 \\ 50 & 275 & 729 \end{bmatrix} \\ &= \frac{l^3}{EI} \begin{bmatrix} 0.33333 & 1.083333 & 2.08333 \\ 5.20833 & 11.4583 & 3.03750 \\ \text{sym.} & & \end{bmatrix} \end{aligned}$$

The stiffness matrix:

$$[K] = [A]^T$$

$$[K] = \frac{EI}{l^3} \begin{bmatrix} 11.0399 & -3.70655 & 0.641026 \\ & 2.37322 & -0.641026 \\ & \text{sym.} & 0.230769 \end{bmatrix}$$

Mass matrix:

$$[M] = \begin{bmatrix} m_1 & 0 & 0 \\ 0 & m_2 & 0 \\ 0 & 0 & m_3 \end{bmatrix} = M \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Natural freq. and mode shape:

$$(-\omega^2 [M] + [K]) \{ \vec{u} \} = 0$$

$$\Rightarrow \left(-m\omega^2 \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} + \left(\frac{EI}{l^3} \right) \begin{bmatrix} x & x & x \\ x & x & x \\ x & x & x \end{bmatrix} \right) \{ u_1, u_2, u_3 \} = 0$$

$$\Rightarrow \left(\begin{bmatrix} x & x & x \\ x & x & x \\ x & x & x \end{bmatrix} - \frac{ml^3}{EI} \omega^2 \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \right) \{ u_1, u_2, u_3 \} = 0$$

$$\text{Define: } \lambda = \frac{ml^3}{EI} \omega^2$$

$$\Rightarrow \left(\begin{bmatrix} x & x & x \\ x & x & x \\ x & x & x \end{bmatrix} - \lambda \begin{bmatrix} 3 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \right) \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = 0$$

$$\Rightarrow \lambda_1 = 0.02502413 ; \quad \omega_1 = \sqrt{EI/ml^3} \cdot \sqrt{\lambda_1}$$

$$\lambda_2 = 0.612216$$

$$\lambda_3 = 4.46008$$

$$\Rightarrow \omega_1 = 0.158224 \sqrt{EI/ml^3}$$

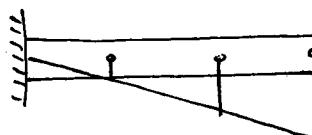
$$\omega_2 = 0.782442 \sqrt{EI/ml^3}$$

$$\omega_3 = 2.11189 \sqrt{EI/ml^3}$$

The modal shapes:

$$\{ \vec{u}_i \} = \left\{ \begin{array}{c} 0.6654970 \\ 0.343615 \\ 0.866597 \end{array} \right\}; \quad \left\{ \begin{array}{c} 0.26073 \\ 0.537842 \\ -0.483283 \end{array} \right\}; \quad \left\{ \begin{array}{c} 0.61646 \\ -0.364393 \\ 0.128367 \end{array} \right\}$$

↪ 0 sign change ↪ 1 sign change ↪ 2 sign change



mode 1



mode 2



mode 3